## FTR - Flight Test Report Dieser Prüfbericht darf ohne schriftliche Zustimmung der EAPR nicht, auch nicht au

Manufacturer	BRUCE GOLDSMITH DESIGN	Type testing No.	
	Bruce Goldsmith Design Hugelweg 12 A-9400 Wolfsberg	serial number	BG021301A
Model	Adam S	Logation	Brauneck
Comment	Schulungstauglich	Location	Achensee

Α



Rev. 2.1 - 06.03.2014 EAPR GmbH - Marktstr. 11 D-87730 Bad Grönenbach - Germany

Date of testing	18.05.2014	Minimum take off v 55 kg	weight	Maximum take off weight 75 kg		
Testpilot		Sepp Bauer		Mike Küng		
Harness		EAPR- Testequipment		EAPR-Testequipment	7	
Pilot's take off weight		60 kg		75 kg	- E	

Classification



Test-criteria		Minimum take off weight	Evaluation	Maximum take off weight	Evaluation
1. Inflation / take-off - 4.1.1					
Rising behavior	ing behavior		А	Smooth, easy and constant rising	А
Special take off technique required		No	A	No	Α
2. Landing - 4.1.2			<u> </u>		
Special landing technique required		No	Α	No	Α
3. Speeds in straight flight - 4.1.3					
Trim speed more than 30km/h		Yes	А	Yes	А
Speed range using the controls larger than 10km/	h	Yes	А	Yes	А
Minimum speed		Less than 25 km/h	А	Less than 25 km/h	Α
4. Control movement - 4.1.4					
Max. weight in flight up to 80kg		Increasing > 55cm	А	Increasing > 55cm	А
Max. weight in flight 80 to 100kg			-		-
Max. weight in flight greater than 100kg			-		-
5. Pitch stability exiting accelerated flight - 4.1	.5	<u> </u>			
Dive forward angle on exit		Dive forward less than 30°	А	Dive forward less than 30°	Α
Collapse occurs		No	Α	No	Α
6. Pitch stability operating controls during acc	elerated f	light - 4.1.6			
Collapse occurs		No	Α	No	Α
7. Roll stability and damping - 4.1.7					
Oscillations		Reducing	А	Reducing	Α
8. Stability in gentle spirals - 4.1.8					
Tendency to return to straight flight		Spontaneous exit	А	Spontaneous exit	Α
9. Behaviour in a steeply banked turn - 4.1.9					
Sink rate after two turns		12m/s to 14m/s	А	12m/s to 14m/s	Α
10. Symmetric front collapse - 4.1.10					
Entry		Rocking back less than 45°	А	Rocking back less than 45°	Α
Recovery	peeds	Spontaneous in less than 3 sec	А	Spontaneous in less than 3 sec	А
Dive forward angle on exit	Ë	0° - 30° Keeping course	Α	0° - 30° Keeping course	Α
Cascade occurs	ŧ	No	A	No	A
Entry	ō	Rocking back less than 45°	А	Rocking back less than 45°	Α
Recovery	accelerated	Spontaneous in less than 3 sec	А	Spontaneous in less than 3 sec	А
Dive forward angle on exit	occ	0° - 30° Keeping course	Α	0° - 30° Entering a turn of less than 90°	Α
Cascade occurs	Ø	No	Α	No	Α

Flight Test Report -Musterprüfnummer: EAPR-GS-0236/14 Seite

Deep stall achieved		Yes				Yes			
Recovery		Spontaneous in less than 3 sec		А	Spontaneous in	less than 3 sec		А	
Dive forward angle on exit		0° - 30°			Α	0° - 30°			Α
Change of course		Changing course less than 45°			Α	Changing course less than 45°			Α
Cascade occurs  12. High angle of attack recovery - 4.1.12		No			А	No			Α
		Chantanagua in loga th	200 2000		^	Spontonoous in	loca than 2 aca		^
Recovery  Cascade occurs		Spontaneous in less th	ian 3 sec		A	Spontaneous in	less than 3 sec		A
13. Recovery from a developed full stall - 4.1.1	3	NO			Α	No			Α
Dive forward angle on exit		0° - 30°			А	0° - 30°			А
Collapse		No collapse			A	A No collapse			A
Cascade occurs (other than collapse)  Rocking backward		No Less than 45°			A	No Less than 45°			A
Line tension		Most lines tight			Α	Most lines tight			Α
14. Asymmetric collapse (trim speed) - 4.1.14		T	1			ı	1		
Change of course until re-inflation	bse	< 90° Dive	e or roll angle	0° - 15°	Α	< 90°	Dive or roll angle	0° - 15°	Α
Re-inflation behavior	trim speed, max 50% collapse	Spontaneous re-inflation	on		Α	Spontaneous re-	inflation		Α
Total change of course  Collapse on the opposite side occurs	rim s 50%	Less than 360°			A	Less than 360° No	A		
Twist occurs	max t	No			A	No No			A
Cascade occurs		No	1		Α	No			Α
Change of course until re-inflation	Se	< 90° Dive	e or roll angle	15° - 45°	Α	90° - 180°	Dive or roll angle	0° - 15°	А
Re-inflation behavior	trim speed, max 75% collapse	Spontaneous re-inflation	on		Α	Spontaneous re-	inflation		Α
Total change of course	trim sp x 75%	Less than 360°			A	Less than 360°			A
Collapse on the opposite side occurs Twist occurs	tri Tax	No No			A	No No			A
Cascade occurs	=	No			A	No			A
Change of course until re-inflation	0	< 90° Dive	e or roll angle	15° - 45°	А	90° - 180°	Dive or roll angle	0° - 15°	А
Re-inflation behavior	accelerated, max 50% collapse	Spontaneous re-inflation	on		A	Spontaneous re-	inflation		A
Total change of course	elera % α	Less than 360°			A	Less than 360°			A
Collapse on the opposite side occurs	acce ix 50	No			A	No			A
Twist occurs	ä	No No			A	No No			A
Cascade occurs  Change of course until re-inflation			e or roll angle	15° - 45°	A A	90° - 180°	Dive or roll angle	0° - 15°	A A
Re-inflation behavior	accelerated, max 75% collapse			10 - 40				0 - 13	
	accelerated, x 75% collap	Spontaneous re-inflation			A	Spontaneous re-inflation  Less than 360°  No  No			A
Total change of course  Collapse on the opposite side occurs	ccel 75%	Less than 360°			A				A A
Twist occurs	ma a	No			Α				Α
Cascade occurs  15. Directional control with a maintained asym	motric col	No			Α	No			А
Able to keep course straight	metric coi	Yes			А	Yes			А
,	180° turn away from the collapsed side possible in 10 sec		Yes			Yes			A
Amount of control range between turn and stall or spin		More than 50% of the symmetric control travel			A	More than 50% of	of the symmetric c	ontrol travel	Α
16. Trim speed spin tendency - 4.1.16			•				<u> </u>		
Spin occurs		No			Α	No			Α
17. Low speed spin tendency - 4.1.17		I.M.							
Spin occurs  18. Recovery from a developed spin - 4.1.18		No			А	No			А
Spin rotation angle after release		Stops spinning in less	than 90°		А	Stops spinning in	n less than 90°		А
Cascade occurs		No			Α	No			A
19. B-line-stall - 4.1.19									
Change of course before release		Changing course less than 45°			Α	Changing course	less than 45°		Α
Behaviour before release		Remains stable with straight span			Α	Remains stable with straight span			Α
Recovery	Recovery		Spontaneous in less than 3 sec			Spontaneous in less than 3 sec			А
Dive forward angle on exit Cascade occurs	•		0° - 30° No			0° - 30° No			A A
20. Big ears - 4.1.20					А				
		Standard technique		А	Special device re	quired		А	
Behaviour during big ears		Stable flight			Α	Stable flight			А
Recovery		Spontaneous in less than 3 sec		Α	Spontaneous in less than 3 sec			Α	
Dive forward angle on exit		0° - 30°			А	0° bis 30°			А
21. Big Ears in accelerated flight - 4.1.21									
Entry procedure		Standard technique			A	Special device re	quired		A
Behaviour during big ears Recovery		Stable flight  Spontaneous in less than 3 sec			A A	Stable flight Spontaneous in	less than 3 sec		A
Dive forward angle on exit	Spontaneous in less than 3 sec  0° - 30°			A	0° bis 30°			A	
Behaviour immediately after releasing the accelarator while		Stable flight			A	Stable flight			A
maintaining big ears  22. Behaviour exiting a steep spiral - 4.1.22						L			
LE. Denavious exiting a steep spiral - 4.1.22									

Flight Test Report - Musterprüfnummer: EAPR-GS-0236/14 Seite

Tendency to return to straight flight	Spontaneous exit	Α	Spontaneous exit	Α
Turn angle to recover normal flight	Less than 720°, spontaneous recovery	Α	Less than 720°, spontaneous recovery	Α
23. Alternative means of directional control - 4.1.2	3			
180° turn achievable in 20 sec	Yes	А	Yes	Α
Stall or spin occurs	No	Α	No	Α
24. Any other flight procedure and/or configuration	n described in the user's manual - 4.1.24			
Procedure works as descibed		NA		NA
Procedure suitable for novice pilots		NA		NA
Cascade occurs		NA		NA
25. Remarks of testpilot:				
	Flugerprobung bei 55kg durch Werkspiloten unter Aufsicht eines	EAPR Testpiloten		
Copyright Ralf Antz 2014 This Flight Test Report was generated automatically and is valid without signature				

Flight Test Report - Musterprüfnummer: EAPR-GS-0236/14 Seite